Origin and Effect of Nonlocality in a Composite

Stewart A. Silling

Prepared by
Sandia National Laboratories
Albuquerque, New Mexico 87185 and Livermore, California 94550

Sandia National Laboratories is a multi-program laboratory managed and operated by Sandia Corporation, a wholly owned subsidiary of Lockheed Martin Corporation, for the U.S. Department of Energy's National Nuclear Security Administration under contract DE-AC04-94AL85000.

Approved for public release; further dissemination unlimited.
Issued by Sandia National Laboratories, operated for the United States Department of Energy by Sandia Corporation.

**NOTICE:** This report was prepared as an account of work sponsored by an agency of the United States Government. Neither the United States Government, nor any agency thereof, nor any of their employees, nor any of their contractors, subcontractors, or their employees, make any warranty, express or implied, or assume any legal liability or responsibility for the accuracy, completeness, or usefulness of any information, apparatus, product, or process disclosed, or represent that its use would not infringe privately owned rights. Reference herein to any specific commercial product, process, or service by trade name, trademark, manufacturer, or otherwise, does not necessarily constitute or imply its endorsement, recommendation, or favoring by the United States Government, any agency thereof, or any of their contractors or subcontractors. The views and opinions expressed herein do not necessarily state or reflect those of the United States Government, any agency thereof, or any of their contractors.

Printed in the United States of America. This report has been reproduced directly from the best available copy.

Available to DOE and DOE contractors from
U.S. Department of Energy
Office of Scientific and Technical Information
P.O. Box 62
Oak Ridge, TN 37831

Telephone: (865) 576-8401
Facsimile: (865) 576-5728
E-Mail: reports@adonis.osti.gov
Online ordering: http://www.osti.gov/bridge

Available to the public from
U.S. Department of Commerce
National Technical Information Service
5285 Port Royal Rd
Springfield, VA 22161

Telephone: (800) 553-6847
Facsimile: (703) 605-6900
E-Mail: orders@ntis.fedworld.gov
Online ordering: http://www.ntis.gov/help/ordermethods.asp?loc=7-4-0#online
Origin and Effect of Nonlocality in a Composite

Stewart A. Silling
Multiscale Science Department
Sandia National Laboratories
P.O. Box 5800
Albuquerque, NM 87185-1322

Abstract

A simple demonstration of nonlocality in a heterogeneous material is presented. By analysis of the microscale deformation of a two-component layered medium, it is shown that nonlocal interactions necessarily appear in a homogenized model of the system. Explicit expressions for the nonlocal forces are determined. The way these nonlocal forces appear in various nonlocal elasticity theories is derived. The length scales that emerge involve the constituent material properties as well as their geometrical dimensions. A peridynamic material model for the smoothed displacement field is derived. It is demonstrated by comparison with experimental data that the incorporation of nonlocality in modeling dramatically improves the prediction of the stress concentration in an open hole tension test on a composite plate.
Contents

1 Introduction ................................................................. 7
2 Microscale model of a composite ........................................ 9
3 Equilibrium of a smoothed displacement field ...................... 12
4 Nonlocal equations of motion .......................................... 15
5 Nonlocality at the macroscale .......................................... 18
6 Conclusions ................................................................. 22

Figures

1 Composite composed of alternating stiff and compliant layers .......... 9
2 Axial displacement fields in the composite layers ....................... 10
3 Strain in the layers for a jump in the smoothed strain ............... 11
4 Stress $\sigma_{yy}$ in a composite panel .................................... 20
5 Midplane stress in a composite panel .................................... 21
1 Introduction

In typical engineering analysis, the elastic response of a heterogeneous material is treated by applying the classical solid mechanics equations with smoothed (often called “homogenized” or “effective”) material properties. These properties can be thought of as those that would be measured using a laboratory test specimen much larger than any internal length scale in the material. For example, we perform an unconfined compression test on a sample of concrete about 6 inches in diameter and measure the total force as a function of displacement at the ends. We divide the relative displacement at the ends by the specimen length and call the result the “strain.” We divide the force by the cross-sectional area and call the result the “stress.” The ratio of stress to strain is defined to be the Young’s modulus of the material, which is then treated as homogeneous for purposes of finite element modeling of a structure.

This approximation is adequate for many applications. However, it ignores the reality of how a load applied on the surface of a concrete body is transmitted internally. Concrete is a heterogeneous material. It consists of small rocks (aggregate) of length scale about 1 inch held together by a weaker material (cement or paste). Since the aggregate inclusions may be in contact with each other, or nearly so, the actual force distribution within the material follows a tortuous path through the aggregate particles and their points of contact. This results in a quantitative and qualitative disparity between the local equations of solid mechanics theory using smoothed material properties and the way the material really behaves. Nonlocality in a random medium such as concrete has been treated in a number of references, for example [38, 16, 15].

In the early 1980’s, Bazant pioneered the application of nonlocal modeling to materials with damage. He demonstrated by a simple example that nonlocality is a necessary property of the elastic response in a material containing distributed defects [3]. In the same general spirit, the present paper derives nonlocal interactions that are implied by the use of a smoothed displacement field to model a heterogeneous microstructure. Unlike Bazant’s analysis, the discussion here omits cracks and damage, and treats only the elastic response of the composite.

Nonlocality also arises in the study of plasticity, in which the finite sizes of dislocations and the distance between them interacts with the geometry of the system, as in the formation of geometrically necessary dislocations. Strain gradient models of plasticity, which incorporate a kind of weak nonlocality, have been developed to model such effects [19, 2]. Much of the nonlocal literature concerns nonlocal operations on a damage variable, particularly its beneficial effects in reducing mesh dependence in numerical modeling [3, 13, 21, 22]. In a heterogeneous elastic material with a periodic microstructure, it has also long been recognized for that nonlocal interactions may arise as a result of homogenization [7, 8, 20, 9, 4, 18, 11, 10]. Ben-Amoz [5, 6] and Ardic, Santare, and Chou [1] incorporated aspects of nonlocality in models of composite material elasticity.

A large body of literature on nonlocal elasticity, much of which is highly mathematical, has been developed over the past five decades. In spite of this, nonlocal models are generally
not adopted in the computational and analytical methods that engineers commonly use for applications. A typical analyst simply does not see why there should be nonlocal forces in a material at any scale above the molecular scale. Casual observers may conclude that in spite of any benefits in regularizing finite element simulations, and regardless of compelling evidence from mathematical proofs, nonlocal models are not justified physically.

What apparently has been lacking in the nonlocal literature is a compelling mechanical picture of nonlocal interactions. In the present work, we consider how nonlocality arises in a specific, relatively simple heterogeneous system, based only on simple mechanical concepts. The micromechanical model uses only the standard equations of solid mechanics, yet it is shown that nonlocality appears in the global model derived from it. It is demonstrated that nonlocality arises from the decision to model the composite in terms of a smoothed displacement field, rather than arising from direct physical interactions across a finite distance. The nonlocal interaction forces can be included in the displacement equations of motion in various nonlocal theories. This example also permits us to compute how mesoscale geometrical dimensions combine with with material properties to determine the length scale that applies in the nonlocal model.
Consider a composite material composed of alternating layers of stiff (s) and compliant (c) materials (Figure 1). The two constituent materials have the same density. The layers have thickness $2h_s$ and $2h_c$. Let $E_s, E_c$ denote the Young’s modulus and $\mu_s, \mu_c$ the shear modulus of each material. It is assumed that

$$E_s \gg E_c, \quad \mu_s \gg \mu_c.$$ 

Only the displacements $u_s$ and $u_c$ in the $x$-direction, that is, parallel to the layers, appear in the following approximate analysis.

**Figure 1.** Composite composed of alternating stiff and compliant layers.

The following analysis is based on a “shear-lag” model of the transfer of forces between the materials [28]. It is assumed that because of the disparity in elastic constants, $u_s$ is independent of $y$. In the compliant material, a displacement field of the form

$$u_c(x, y) = u_s(x) + (h_c^2 - y^2)w(x)$$

is assumed, where $w$ is a continuous function, and $y = 0$ is located at the midplane of a typical compliant layer (Figure 2). The shear traction on either surface of any compliant layer is found from

$$\tau(x) = \mu_c \frac{\partial u_c}{\partial y}(x, -h_c) = 2\mu_c h_c w(x).$$
In the absence of body forces, a force balance on the cross-section of a stiff layer yields

\[ h_s \sigma_s'(x) + \tau(x) = 0 \]  \hspace{1cm} (3)

where \( \sigma_s \) is the normal stress and prime denotes \( d/dx \). Using (2), (3), and the Hooke’s law expression

\[ \sigma_s = E_s u'_s, \]

the force balance on the stiff layers may be rewritten as

\[ E_s h_s u''_s(x) + 2 \mu_c h_c w(x) = 0. \]  \hspace{1cm} (4)

**Figure 2.** Axial displacement fields in the composite layers.
Figure 3. Strain fields in the stiff and compliant layers resulting from a prescribed smoothed displacement field in which there is a jump in strain at $x = 0$. 

Strain 
\[ \varepsilon(x) \] 
Smoothed strain \( \varepsilon'(x) \) 
Stiff strain \( \varepsilon'_s(x) \) 
Compliant strain \( \varepsilon'_c(x) + 2h_c w'(x)/3 \)
3 Equilibrium of a smoothed displacement field

Define the smoothed (homogenized) displacement field to be the average at any \( x \) over a cross-section through the layers:

\[
\bar{u}(x) = \frac{1}{h_s + h_c} \left[ h_s u_s(x) + \int_{0}^{h_c} u_c(x, y) \, dy \right].
\]

Using (1), this evaluates to

\[
\bar{u}(x) = u_s(x) + \alpha w(x),
\]

where

\[
\alpha = \frac{2h_c^3}{3(h_s + h_c)}.
\]

Using (5) and (6), we remove \( u_s \) from the force balance (4) and obtain

\[
\bar{u}''(x) = \alpha w''(x) - \frac{2\mu_c h_c}{E_s h_s} w(x).
\]

Our immediate objective is to analyze the forces in the composite for a prescribed \( \bar{u} \) function. If \( \bar{u} \) is prescribed, then (7) is a nonhomogeneous second order linear ordinary differential equation for \( w \). The homogeneous part of the solution to this ODE (that is, for \( \bar{u} \) either constant or a linear function of \( x \)) is

\[
w_h(x) = Ae^{\lambda x} + Be^{-\lambda x},
\]

where

\[
\lambda = \sqrt{\frac{2\mu_c h_c}{\alpha E_s h_s}} = \sqrt{\frac{3\mu_c (h_s + h_c)}{E_s h_s h_c^2}}
\]

and where \( A \) and \( B \) are arbitrary constants.

Of particular interest is the choice of prescribed \( \bar{u}'' \) given by

\[
\bar{u}''(x) = \Delta(x)
\]

where \( \Delta \) is the Dirac delta function. Kinematically, this choice means that the homogenized strain field \( \bar{u}' \) is given by the Heaviside step function,

\[
\bar{u}'(x) = H(x)
\]

(Figure 3). For this choice, the solution, denoted \(-G\), to the ODE (7) is

\[
w(x) = -G(x), \quad G(x) = ke^{-\lambda|x|},
\]

where

\[
k = \frac{1}{2\alpha \lambda}.
\]
The easiest way to confirm that (10) and (11) provide a solution to (7) and (9) is to observe that by integrating (7) from $0^-$ to $0^+$, the jump in $\bar{u}'$ at the origin is given by

$$[\bar{u}'] = \alpha [w'] - \frac{2 \mu_c h_c}{E_s h_s} \int_{0^-}^{0^+} w(x) \, dx. \quad (12)$$

By (10), $w$ is bounded (as well as continuous) everywhere, so the last term vanishes. Differentiation of $w$ yields

$$w'(x) = k\lambda (2H(x) - 1)e^{-\lambda|x|}.$$ 

Evidently this implies $[w'] = 2k\lambda$. So, using (11), the jump condition (12) is satisfied.

Because the ODE (7) is linear, and because of the properties of the Dirac delta function, for an arbitrary prescribed function $\bar{u}$, (10) implies

$$w(x) = -\int_{-\infty}^{\infty} \bar{u}''(p)G(x - p) \, dp. \quad (13)$$

Next we compute the homogenized normal stress $\bar{\sigma}$ in the composite. From a force balance on a cross-section of through the point $x$, this stress is found to be

$$\bar{\sigma}(x) = \frac{1}{h_s + h_c} \left[ h_s \sigma_s(x) + \int_0^{hc} \sigma_c(x, y) \, dy \right] \quad (14)$$

with

$$\sigma_s(x) = E_s u_s'(x), \quad \sigma_c(x) = E_c u_c'(x).$$

Carrying out the integration in (14) using (1) and (5) leads to

$$\bar{\sigma}(x) = E \bar{u}'(x) - \gamma w'(x) \quad (15)$$

where

$$\bar{E} = \frac{E_s h_s + E_c h_c}{h_s + h_c}, \quad \gamma = \alpha \bar{E} - \frac{2E_c h_c^3}{3(h_s + h_c)} = \frac{2h_s h_c^3 (E_s - E_c)}{3(h_s + h_c)^2}. \quad (16)$$

Rewriting (15) using (13) and (17) leads to

$$\bar{\sigma}(x) = \bar{E} \bar{u}'(x) + \gamma \int_{-\infty}^{\infty} \bar{u}''(p)G'(x - p) \, dp. \quad (18)$$

Note that $w$ no longer appears explicitly in this expression for the homogenized stress. The nonlocality inherent in modeling heterogeneous materials using a smoothed displacement field is now evident: values of $\bar{u}$ remote from $x$ contribute to the stress at $x$. This is a type of strong nonlocality. A more suggestive version of (18) may be obtained by integrating the last term by parts:

$$\bar{\sigma}(x) = \bar{E} \bar{u}'(x) + \gamma \int_{-\infty}^{\infty} \bar{u}'(p)G''(x - p) \, dp. \quad (19)$$
Differentiating (10) twice yields

\begin{align*}
G''(x) &= k\lambda^2 e^{-\lambda|x|} - 2k\lambda\Delta(x) \\
&= \lambda^2 G(x) - 2k\lambda\Delta(x). \tag{20}
\end{align*}

Substituting (21) into (19), using the properties of the Dirac delta function, and using (11), (16), and (17) to simplify the constants yields

\begin{equation}
\bar{\sigma}(x) = E_c \bar{u}'(x) + \gamma\lambda^2 \int_{-\infty}^{\infty} \bar{u}'(p)G(x - p) \, dp. \tag{22}
\end{equation}

This expression is similar to the nonlocal stress used in Eringen’s nonlocal theory [17], because it involves taking a weighted average of strain.
4 Nonlocal equations of motion

Next we investigate how the nonlocal forces in the homogenized model appear in nonlocal equations of motion. With this goal in mind, we evaluate the acceleration at \( x \) at a given time \( t \). From the linear momentum balance in the absence of body forces, using the assumption that \( \rho_s = \rho_c \), our expression for the net normal stress (22) implies

\[
\rho \ddot{u}(x) = \sigma'(x) = E_c \ddot{u}(x) + \gamma \lambda^2 \int_{-\infty}^{\infty} \ddot{u}(p)G'(x-p) \, dp.
\]

Integrating the last term by parts,

\[
\rho \ddot{u}(x) = E_c \ddot{u}(x) + \gamma \lambda \int_{-\infty}^{\infty} \ddot{u}(p)G''(x-p) \, dp.
\]

Using (20) in the last term, and introducing a body force field \( b \) according to d’Alembert’s principle, yields

\[
\rho \ddot{u}(x) = E_c \ddot{u}(x) + \gamma k \lambda^4 \int_{-\infty}^{\infty} (\ddot{u}(p) - \ddot{u}(x)) e^{-\lambda|x-p|} \, dp + b(x) \quad (23)
\]

In this form, the model is similar to the nonlocal theory proposed by DiPaola, Failla, and Zingales [14]. This form, like Kröner’s [24], retains both local and nonlocal terms.

The peridynamic model seeks to eliminate any reference to strain, because one of its goals is to avoid using spatial derivatives of the deformation, thus making the theory compatible with discontinuities in displacement. The general form of the peridynamic equation of motion in one dimension [31, 34] is given by

\[
\rho \ddot{u}(x) = \int_{\mathcal{H}_x} f(p,x,t) \, dp + b(x) \quad (24)
\]

where \( \mathcal{H}_x \) is a neighborhood of \( x \) called the family of \( x \). The radius of \( \mathcal{H}_x \), which is called the horizon, can be either finite or infinite. The function \( f \) is called the bond force density. It can depend quite generally on the deformation of the family through suitable material models, including the effects of nonlinearity and damage. The peridynamic model is nonlocal because the family has nonzero size.

In linearized peridynamics [32], the equation of motion (24) can be approximated by the following expression, which is formally the same as in the nonlocal theories of Kunin [25, 26] and Rogula [30]:

\[
\rho \ddot{u}(x) = \int_{\mathcal{H}_x} C(x,p)(u(p) - u(x)) \, dp + b(x) \quad (25)
\]

where \( C \) is a function called the micromodulus. The equation of motion (23) that was derived from the microstructural model of a composite has strong similarities to the linearized peridynamic expression (25). Both equations contain strongly nonlocal terms; in (23) the horizon is infinite, although a reasonable approximation would be to cut off the nonlocal interactions outside a distance \( r \) where the weighting term \( e^{-\lambda r} \) is sufficiently small.
A local term involving \( u'' \) does not appear in the peridynamic equation (25). However, with the goal of representing this term in (23), it can be approximated by short-range interactions by using the same sort of manipulations used above. Using integration by parts, the following identities hold:

\[
\begin{align*}
u''(x) &= \int_{-\infty}^{\infty} u''(p) \Delta(x - p) \, dp \\
&= \int_{-\infty}^{\infty} u'(p) \Delta'(x - p) \, dp \\
&= \int_{-\infty}^{\infty} u(p) \Delta''(x - p) \, dp.
\end{align*}
\]  

(26)

Using the approximation

\[
\Delta(x) \approx \phi(x) := \frac{\tau e^{-\tau|x|}}{2},
\]

where \( \tau \) is a large constant, we compute from (26)

\[
\begin{align*}
\int_{-\infty}^{\infty} u(p) \Delta''(x - p) \, dp &\approx \int_{-\infty}^{\infty} u(p) \phi''(x - p) \, dp \\
&= \int_{-\infty}^{\infty} u(p) \left( \frac{\tau^3 e^{-\tau|x-p|}}{2} - \tau^2 \Delta(x - p) \right) \, dp \\
&= \frac{\tau^3}{2} \int_{-\infty}^{\infty} (u(p) - u(x)) e^{-\tau|x-p|} \, dp.
\end{align*}
\]

Thus, our model for the smoothed composite displacement field, (23), can be approximated to any level of accuracy by choosing sufficiently large \( \tau \) in the following model:

\[
\rho \ddot{\bar{u}}(x) = \int_{\mathcal{H}_x} \left( \frac{E c \tau^3}{2} e^{-\tau|x-p|} + \gamma k \lambda^4 e^{-\lambda|x-p|} \right) \left( \bar{u}(p) - \bar{u}(x) \right) \, dp + b(x).
\]

(27)

This is the peridynamic model (25) with the choice of micromodulus function given by

\[
C(x, p) = \frac{E c \tau^3}{2} e^{-\tau|x-p|} + \gamma k \lambda^4 e^{-\lambda|x-p|}.
\]

The foregoing discussion illustrates many of the key properties in understanding the role of nonlocality in the modeling of continua:

- Nonlocality is introduced by making the modeling decision to describe the problem using a smoothed displacement field rather than the detailed microstructural fields.
• Although the underlying microstructural model (in terms of the $u_s$ and $u_c$ fields) is local, the resulting smoothed model (in terms of $\bar{u}$) is nonlocal.

• Expressions called bond force densities in the peridynamic model of the form 
  \[ C(x, p)(\bar{u}(p) - \bar{u}(x)) \]
  do not necessarily represent a direct nonlocal physical interaction (such as electrostatic forces) between $p$ and $x$.

• The applicable length scale in the nonlocal term, $1/\lambda$, depends not only on the geometrical length scales ($h_s$ and $h_c$), but also on the constituent material properties $E_s$ and $\mu_c$. If $E_s \gg \mu_c$, then this length scale can greatly exceed the layer thicknesses. This result is consistent with the computations by Pipes and Pagano [29] which show that edge effects on the stresses in plies can extend over distances that substantially exceed the ply thicknesses.

• A peridynamic model can contain multiple length scales; in the case of the composite model in (27) these are $1/\lambda$ and $1/\tau$.

• By considering a displacement field of the form 
  \[ \bar{u}(x) = \varepsilon_0 x + \frac{\beta}{2} x^2, \]
  where $\beta$ is a constant, and applying (23), one finds that the quadratic term leads to an acceleration at $x = 0$ given by
  \[ \ddot{\bar{u}}(0) = \frac{4\beta\gamma k\lambda}{\rho}. \]
  Since this acceleration is positive whenever the strain gradient $\beta$ is positive, this result means that positive strain gradient tends to increase the force on $x$. This is suggestive of experimental results that show stresses in real materials increase as the strain gradient is increased [19].

The analogous local model for the composite in terms of the smoothed displacement field is

\[ \rho\ddot{\bar{u}}(x) = \bar{E}\dddot{\bar{u}}(x) + b(x) \]

where $\bar{E}$ is given by (16). Comparing this with (23), evidently the local term $\bar{E}\dddot{\bar{u}}$ is replaced by a different local term $E_c\dddot{\bar{u}}$ plus a nonlocal term. It is possible that a more detailed microstructural model, for example a sub-microstructural model that considers the features within each of the layers such as individual fibers, would further resolve the local term in (23). Based on the patterns emerging in the above derivations, it plausible that a hierarchy of $N$ such sub-models could be derived, resulting in a micromodulus of the form

\[ C(x, p) = C_1(x, p; \lambda_1) + C_2(x, p; \lambda_2) + \cdots + C_N(x, p; \lambda_N) \]

where each $C_i$ represents interactions with length scale $1/\lambda_i$. By inference from (27), such terms might have the form

\[ C_i(x, p; \lambda_i) = a_i e^{\lambda_i|x-p|} \]

where the $a_i$ are constants.
5 Nonlocality at the macroscale

As noted in the previous section, the operative length scale in a heterogeneous material system depends on the constituent material properties as well as the geometrical length scale. However, the macroscale geometry of a body also combines with material properties to provide additional length scales. For example, consider the classic problem of an anisotropic plate under tension containing an open hole of radius \( r \) (Figure 4). In the local theory, the stress \( \sigma_{yy} \) as a function of position \( x \) along the midplane was derived analytically by Leknitskii [27]:

\[
\sigma_{yy} = \frac{\sigma_\infty}{2} \left\{ 2 + \left( \frac{r}{x} \right)^2 + 3 \left( \frac{r}{x} \right)^4 + (2 - n) \left[ 5 \left( \frac{r}{x} \right)^6 - 7 \left( \frac{r}{x} \right)^8 \right] \right\}
\]

(28)

where

\[
n = \sqrt{2 \left( \frac{E_1}{E_2} - \nu_{12} \right)} + \frac{E_1}{G_{12}}\]

where the subscripts 1 and 2 denote the loading direction and transverse direction respectively. From this, the stress concentration at the edge of the hole is found to be

\[
K_{\pi/2} = 1 + n.
\]

The parameter \( n \) tends to increase in anisotropic materials, particularly those that have greater stiffness in the loading (\( y \)) direction. In addition to changing the stress concentration at the edge, anisotropy also changes the rate at which the stress decays with distance from the edge. Denote by \( r_0 \) the radius at which the stress in the classical solution drops off to half of its value at the edge, that is,

\[
\frac{\sigma_{yy}(r_0)}{\sigma_{yy}(r)} = \frac{1}{2}.
\]

Typical values of \( r_0 \) determined from (28) are given in Table 1. (Similar decay distance parameters play an important role in certain laminate failure models that implicitly recognize nonlocality [37, 23] by including an explicit length scale.)

<table>
<thead>
<tr>
<th>Material</th>
<th>( n )</th>
<th>( K_{\pi/2} )</th>
<th>( r_0/r )</th>
</tr>
</thead>
<tbody>
<tr>
<td>Isotropic</td>
<td>2</td>
<td>3</td>
<td>1.52</td>
</tr>
<tr>
<td>Fabric ply</td>
<td>4.16</td>
<td>5.16</td>
<td>1.13</td>
</tr>
<tr>
<td>Unidirectional ply</td>
<td>10</td>
<td>11</td>
<td>1.07</td>
</tr>
</tbody>
</table>

Table 1. Stress concentration and decay distance near an open hole in a plate.

Thus, in a unidirectional ply, the stress decays to half its value at the edge over a distance of only 7% of the hole radius. If the hole radius is \( r = 2.5 \text{mm} \), then this distance is \( r_0 - r = 0.175 \text{mm} \), which is on the order of a typical ply thickness.
To put this result in the context of our previous discussion of length scales and nonlocality in heterogeneous media, compute the length scale implied by \( \lambda \) with 
\[
h_c = h_s = 0.1 \text{mm}, \quad \mu_c = 4 \text{GPa}, \quad E_s = 150 \text{GPa}.
\]
The resulting length scale for nonlocal interactions due to transfer of shear load between the constituent materials is
\[
\frac{1}{\lambda} = \sqrt{\frac{E_s h_s h_c^2}{3 \mu_c (h_s + h_c)}} = 0.25 \text{mm}.
\]
Comparing this with our stress concentration decay distance (0.175mm), evidently nonlocality in a homogenized model due to the exchange of forces between the materials could play an important role in modeling the problem.

Toubal, Karama, and Lorrain [36] measured the stress in a fabric composite laminate near an open hole as a function of position along the midplane of the specimen. They used an electronic speckle pattern interferometry (ESPI) measurement technique. This technique provides non-contact strain data with a spatial resolution of about 0.5mm. Their measurements showed that the actual stress concentration is much lower than what is predicted by the analytical results from the local theory [27]. Does nonlocality explain this difference?

To investigate this possibility, the peridynamic computational model Emu [33] was applied to try to reproduce the measured stress concentration reported in [36]. The material model used in the peridynamic computations was similar to that used in [39]. In this material model, peridynamic bonds parallel to the fibers have much greater stiffness than bonds in any other direction.

In the experiment, the hole diameter was 5.0mm, and the specimen width was 25mm. The specimen contained 6 plies, all epoxy reinforced by carbon fabric, with a total thickness of 2.28mm. The ply laminate properties were \( E_1=51 \text{GPa}, \ E_2=50 \text{GPa}, \ \nu_{12}=0.06, \ G_{12}=3.24 \text{GPa} \). Since all the plies were identical in the experiment, shear forces parallel to the plies are not significantly involved in the problem. Therefore, for purposes of estimating the nonlocal interaction distance \( 1/\lambda \), the applicable geometric length scale is the fabric tow width. In other words, each tow acts like one of the layers in the microstructural model developed in Section 2. This view is supported by the X-ray diffraction studies by Davies et al. [12] which show that in the vicinity of an open hole, the fabric tows deform more or less uniformly across the width of each tow. Under this assumption, and setting \( h_c = h_s = 1.25 \text{mm} \), which is a typical value for tow width, one finds from (8) that \( 1/\lambda = 2.0 \text{mm} \). In the computational model, mesh spacing was 0.32mm, and the peridynamic horizon was 2.0mm.

As shown in Figure 4, the predicted contours of \( \sigma_{yy} \) (normal stress in the loading direction) show strong gradients in the vicinity of the hole. In fact, there are large gradients above and below the hole as well, due primarily to the relatively small shear modulus \( G_{12} \), which is characteristic of fiber and fabric reinforced composites. In an isotropic material such as a typical metal, the contours of stress would be more diffuse. Figure 5 shows a comparison between the optically measured stress \( \sigma_{yy} \) and the local theory of anisotropic media [36]. The figure also shows the results from the peridynamic computational model. The results in this figure suggest that nonlocality helps improve the agreement between a continuum model and
measured data for stress concentrations in composites. This also helps explain why failure
criteria in composites that rely solely on the predicted stress in the local theory tend to
under-predict the failure load in open-hole tension and compression tests: these criteria are
based on an over-prediction of the stress concentration [35].

Figure 4. Stress contours in a simulated open hole tension
test in a composite. Note the large stress gradients.
Figure 5. Comparison of stress along the midplane in an open hole tension test on a fabric-reinforced composite. The local theory overpredicts the stress concentration compared with optically measured data [36]. The peridynamic model offers improved agreement, apparently due to nonlocality.
6 Conclusions

The purpose of this work is to show how a nonlocal model arises when we make the modeling decision to use a smoothed displacement field, rather than a detailed microstructural description. By considering the micromechanics of a layered composite under uniaxial stress, it was shown that nonuniformity of the displacement field across any cross-section leads to nonlocality in a homogenized model. The nonlocal effects appear only when the strain in the smoothed displacement field is non-constant (that is, when a strain gradient is present). The nonlocal interactions, in this special case, can be represented using different nonlocal models, including those of Eringen and Kröner, as well as peridynamics. The peridynamic micromodulus function for the nonlocal interactions can be determined explicitly, although there is some arbitrariness in the kernel used to approximate the local term that appears in the homogenized equation of motion. The length scale $1/\lambda$ in the peridynamic model is determined not only by the microscale geometry of the composite, but also by the material properties of the constituent materials. In the peridynamic expression that was derived, interactions between material points separated by a finite distance necessarily occurs, even though there is no direct physical interaction between these points in the microstructure.

In summary, nonlocality is not just a property of the physical system; it is also a property of the fields we choose to model the system with.
References


DISTRIBUTION:

<table>
<thead>
<tr>
<th>Quantity</th>
<th>Code</th>
<th>Description</th>
</tr>
</thead>
<tbody>
<tr>
<td>1</td>
<td>MS 0899</td>
<td>Technical Library, 9536 (electronic copy)</td>
</tr>
<tr>
<td>1</td>
<td>MS 0359</td>
<td>D. Chavez, LDRD Office, 1911</td>
</tr>
</tbody>
</table>